STEREO IMPACT

PROBLEM REPORT PR-7005 SEPT-Counting 2004-10-10

PR Numbers: 1xxx=UCB, 2xxx=Caltech/JPL, 3xxx=UMd, 4xxx=GSFC/SEP, 5xxx=GSFC/Mag, 6xxx=CESR, 7xxx=Kiel, 8xxx=ESTEC, 9xxx=MPAe					
Assembly : IMPACT SEPT-NS FM1		SubAssembly : Electronics Box			
Component/Part Number: FPGA/RAM		Serial Number: A201 SN2			
Originator: Reinhold Mueller-Mellin		Organization: U. Kiel			
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Failure Occurred During (Check one $$)					
Functional test	Qualification test	□ S/C Integration	□ Launch operations		
Environment when failure occurred:					
√ Ambient	\Box Vibration				
□ Thermal	□ Vacuum	□ Thermal-Vacuum	□ EMI/EMC		
Γ					
Problem Description					
On October 10, 2004, FM1 SEPT (both –E and –NS) were connected to FM1 SEP-Central and FM1 LVPS for the first time. Power consumption and a run in nominal observation mode were flawless, however, a run					
in test mode showed missing counts in two out of four histograms (dubbed PDFE 0 and 1) of SEPT-NS.					
When in test mode, an in-flight test pulse generator is running at 17.5 kHz. For an accumulation period of					
	60 seconds, the expected counting result is 1,047,552. This was shown by PDFE 2 and 3, but PDFE 0 and 1				
showed a factor ~10 less. This failure may have been present from the time of assembly of the electronics,					
but went undetected, because the test pulse runs were shortened to one second to save time. Counts are not					
missing at low countin	g results (e.g. at 17,500), but o				
Analyses Performed to Determine Cause					
SEPT-NS was connected to the GSE and by varying the accumulation time, the erroneous behaviour could					
be tracked down to a stuck bit 15 in the 24 bit counting arithmetic. Possible failure sources are the Actel					
FPGA and the Static RAM. The FPGA was burned with the "old" algorithm. As this failure case does not invalidate the EMC test, SEPT-NS will continue with EMC. All other units (FM1 SEPT-E and FM2 SEPT-					
	ambient and do not show the fa				
	v. They point to a stuck bit in the				
		tion/ Resolution			
□ Rework	√ Repair	Use As Is	□ Scrap		
A quick repair is possi	ble by exchanging the digital e	lectronics board in SEPT			
board in the flight spare (A195/201 S/N5). (see PR-7004 SEPT-Accident). This can be achieved in time					
before the scheduled vibration and TV tests. The FPGA (RT54SX32S-CQ208B, LDC0202 5,					
S/NT25JS001) will be replaced by a spare part burned with the new algorithm, and the repaired digital					
board (DB-A1-2G) tog	gether with the analog board (A	(B-A1-2D) will serve in t	he flight spare.		
A review of the Actel	design and board layout has de	etermined that there were	no issues with the design or		
A review of the Actel design and board layout has determined that there were no issues with the design or layout. This review was performed by the Office of Logic Design. It is not certain whether or not the failure					
is due to an antifuse failure, but it does appear that it is something inside the part. There is nothing within					
the design or layout that contributed to the failure. This part will be submitted to Actel for failure analysis					
through the University	of Kiel. The results of this ana	lysis is not expected to b	e available until after launch.		
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This problem has not been seen on any of the other 3 flight units nor has it been seen through recent tests on this unit with the newly programmed part.					
Date Action Taken: NOV-04 Retest Results : Board level test passed (15-NOV-04) and					
has successfully ompleted all instrument level environmental tests.					
	Corrective Action Required on other Units \Box Serial Number(s): n/a				
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Date:

Closure Approvals				
Subsystem Lead: IMPACT Project Manager:	Reinhold Mueller-Mellin	Date: 08-DEC-04 Date:		
IMPACT QA:		Date:		

The following tests were performed in order to identify the origin of the problem:

NASA IMPACT Instrument Manager:

FM1 SEPT-NS was connected to the ESTEC GSE. The first test was conducted by using different accumulation times (0.01, 0.1, 1, 2, 3, 4, 300, 600 and 800 seconds) with the on-board test pulse generator producing pulses at 17.579 kHz. The results show that for PDFE0 and PDFE1 channels, the recorded numbers of counts are wrong for t > 2 seconds. In fact, the recorded numbers of counts were the expected one modulo 32768 (with marginal noise on top). For PDFE2 and PDFE3 (as well as for all the other three units), the recorded numbers of pulses are exactly the expected ones. A second test was performed in order to see whether 32768 counts could be recorded. By tuning the different parameters such as accumulation time and gain of PDFE0, it was possible to obtain operating conditions for which the expected number of counts should statistically be 32768. The test was performed 1000 times to be statistically valid. The values 32766, 32767 as well as 1, 2 were obtained but not 32768, instead the value 0 was recorded. PDFE1 followed the same behaviour.

The energy histogram counters (32 bins in the present mode) of each channel are 24 bit wide (stored in 32bit-word-aligned address in the SRAM). The SRAM is addressed byte-wise, consequently for each counter access, 3 byte reads are performed, followed by three byte writes. PDFE0 and PDFE1 channels share the same SRAM interface pins as well as the same FPGA implementation of the 24-bit adder performing the increment. The missing value 32768 strongly suggests that the problem is internal to the FPGA. Indeed, a hypothetical problem in the SRAM would appear on the 3 bytes of each histogram counter, however, the observed problem is independent of the SRAM address. Since the value 32767 can be read from the SRAM, the problem is very likely located in the adder where bit 15 could be stuck.

Rework completed (November 2004):

Both electronics boards (analog and digital) were disassembled from the flight spare unit and assembled into the SEPT-NS FM1 unit at ESTEC from 05-NOV-04 through 12-NOV-04. The digital board is equipped with a FPGA already burned with the new algorithm as a result of the SEPT accident (see PR-7004 SEPT-Accident). The failed digital board has received a new SRAM and was still failing pointing unambiguously to the FPGA as the culprit. The board will receive a new FPGA, also burned with the new algorithm, and together with its analog board serve as new flight spare electronics.

Retest completed (November 2004):

On 15-NOV-04 SEPT-NS FM1 was tested thoroughly with special attention to the counting arithmetic. It showed nominal performance and was declared ready for the repetition of the environmental tests. The new flight spare electronics is still awaiting rework and check-out.

Rework of Flight Spare completed (June 2005):

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The stressed FPGA was replaced by a spare part burned with the new algorithm. The repaired digital board was installed into the E-box of SEPT Flight Spare, tested with one temperature cycle. The test was successful.