STEREO IMPACT

🗆 Vacuum

□ Thermal

□ EMI/EMC

PR Numbers: 1xxx=UCB, 2xxx=Caltech/JPL, 3xxx=UMd, 4xxx=GSFC/SEP, 5xxx=GSFC/Mag, 6xxx=CESR, 7xxx=Keil, 8xxx=ESTEC, 9xxx=MPAe						
Assembly : SIT Instrument		SubAssembly :				
Component/Part Number:		Serial Number: 01				
Originator: Walpole		Organization: UMd				
Phone : 301-405-6217		Email : Walpole@sampex.umd.edu				
Failure Occurred □ Functional test	During (Check one $$) x Qualification test	□ S/C Integration	□ Launch operations			
Environment when failure occurred:						
□ Ambient	\Box Vibration					

Problem Description

x Thermal-Vacuum

During thermal balance testing, a check of the wiring of the SIT Acoustic Door actuator showed that from the outside of the chamber, the wiring was incorrect. We expected to see about 5 ohms between pins 1 and 6 and between 2 and 7 per table 4.8.3 below taken from the Harness Specification document. We expected pins 1 and 2 to be shorted to each other. Likewise pins 6 and 7.

What we found was 5.5 ohms between pins 1 and 2 and between 6 and 7. Pins 1 and 6 were shorted together, as were pins 2 and 7.

Analyses Performed to Determine Cause

We consulted the technician who wired the SIT actuator onto J3 on the SIT box. She confirmed that the wiring error was on SIT and was the result of misunderstanding the wording on the Table.

Corrective Action/ Resolution				
x Rework	Repair	□ Use As Is	□ Scrap	

To allow the SIT door to be opened during TB, a 9 pin - 9 pin adaptor was made correcting the pinout. SIT FM1 harness has been corrected and verified according to the updated drawing. To prevent the problem from occurring on SIT FM2 we have supplied a clarifying schematic (below).

 Date Action Taken:
 3/10/05
 Retest Results:
 Harness verified on both units

 Corrective Action Required/Performed on other Units
 Serial Number(s):
 n/a

Closure Approvals Subsystem Lead: Date: IMPACT Project Manager: Date IMPACT QA: Date: NASA IMPACT Instrument Manager: Date:

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PROBLEM REPORT PR-3007 Walpole 2/18/05

(From IMPACTHarnessSpec_J.doc Version J – 2004-Mar-24)

4.8.3. SIT-P3 (S/C to SIT)

<u>Pin</u>	Signal	Destination	Harness
1	SIT ACT	Spacecraft	#20 TSP w/6
2	SIT ACT	Spacecraft	#20 TSP w/7
3	Spare		
4	Spare		
5	Spare		
6	SIT ACT RTN	Spacecraft	#20 TSP w/1
7	SIT ACT RTN	Spacecraft	#20 TSP w/2
8	Spare		
9	SITchassis ground		

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