

## **Agenda**

**8:30-12:00 Status reports**

**STEREO Project Status- Project (20 min)**

**IMPACT Overall Status- Luhmann (10 min)**

**BOOM Package:**

**MAG- Acuna (15 min)**

**SWEA- Aoustin/Sauvaud (15 min)**

**STE- McBride (15 min)**

**Boom- Ullrich (15 min)**

**Break (10 min)**

**SEP Package:**

**SEPT- Mueller-Mellin (15 min)**

**LET/Central- Mewaldt (15 min)**

**HET/Mechanical- von Roseninge (15 min)**

**SIT- von Roseninge (15 min)**

**IDPU/GSE - Curtis (15 min)**

**LVPS/SIT HVPS - Berg (15 min)**

**12:00-1:00 Lunch, Discussion**

**1:00-3:00**

**IDPU Software Requirements / Design review-  
Curtis (2 hours)**

**Break (10 min)**

**3:10-6:00**

**Issues: Curtis/all**

**PAIP- (20 min)**

**ICD- (10 min)**

**EMC/Grounding- (20 min)**

**Telemetry/Beacon- (20 min)**

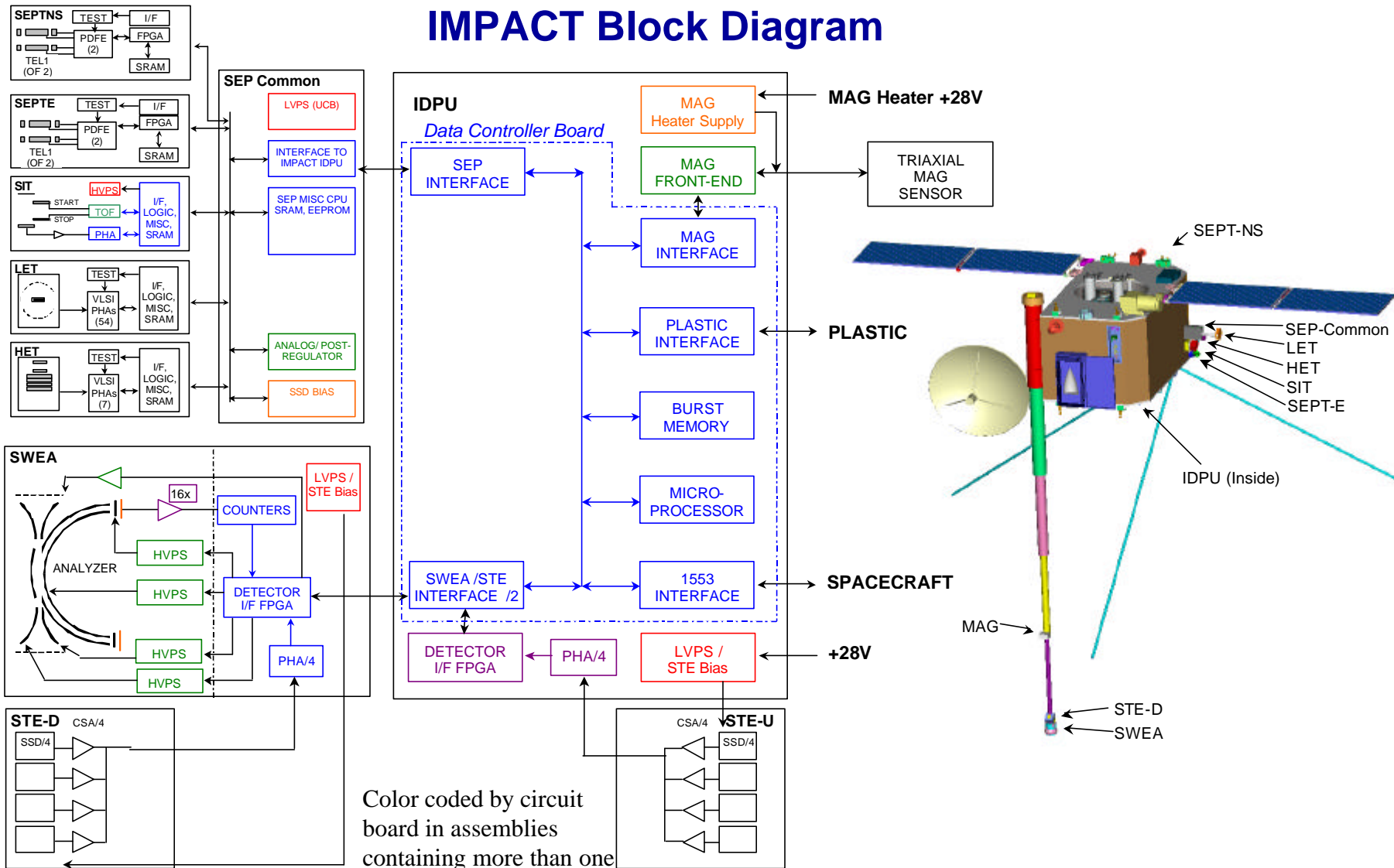
**Contamination Control- (20 min)**

**Environmental Requirements- (20 min)**

**Schedule- (20 min)**

## **IDPU Status**

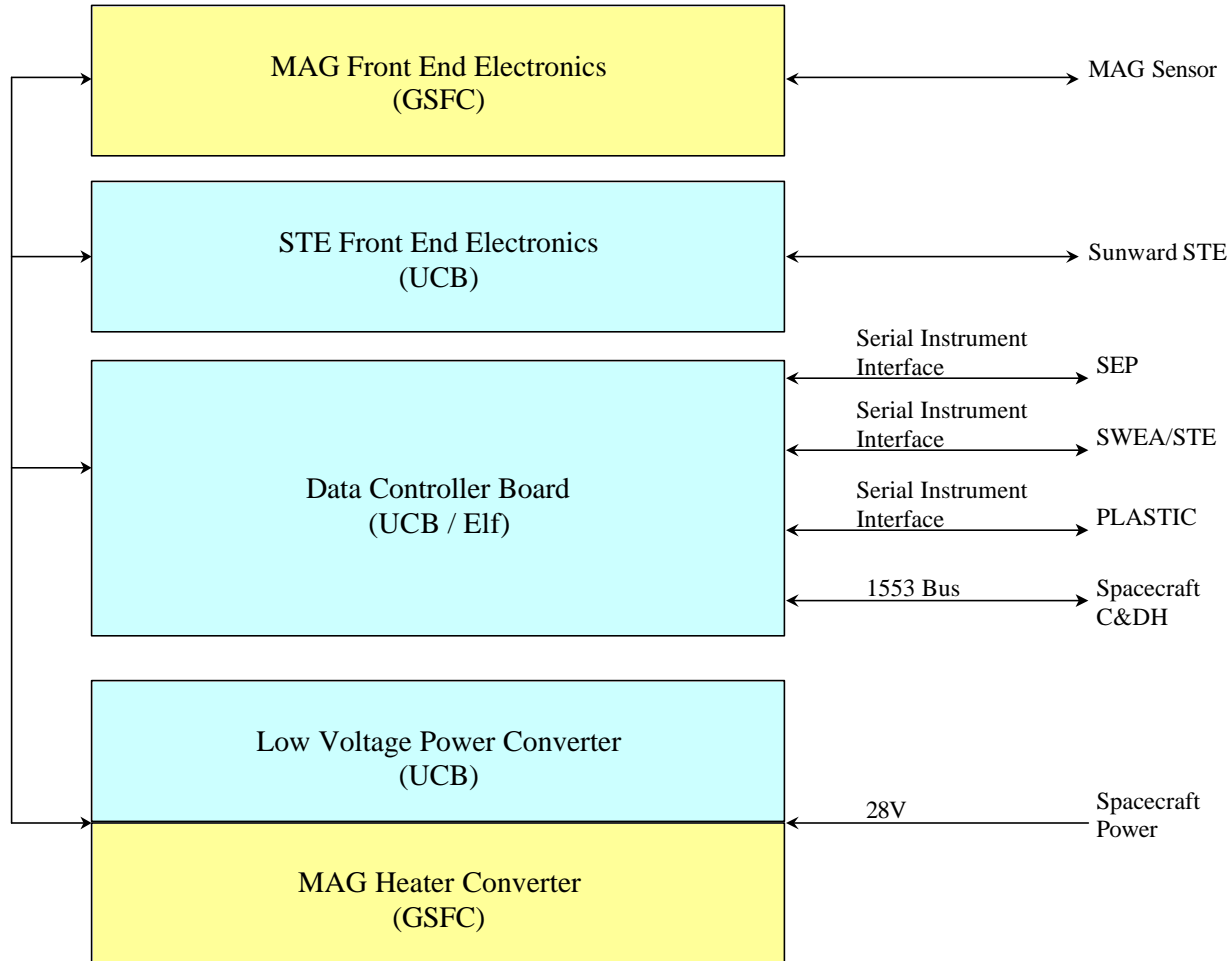
## IMPACT Block Diagram



Color coded by circuit board in assemblies containing more than one board.

"Stackable Connector" Interconnect:  
+/-12V, +/-5, +5V, +2.5V  
MAG Serial Instrument Interface  
MAG Heater Control  
STE Serial Instrument Interface

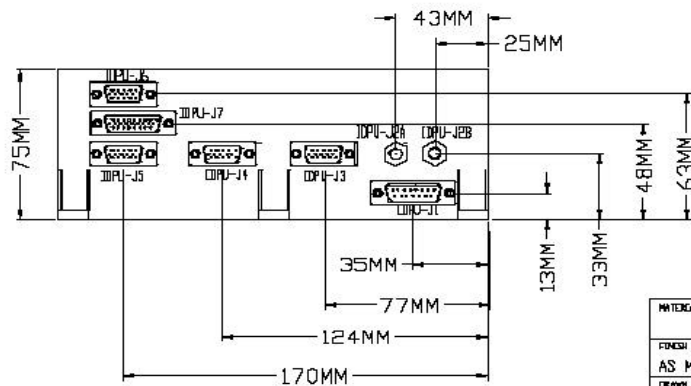
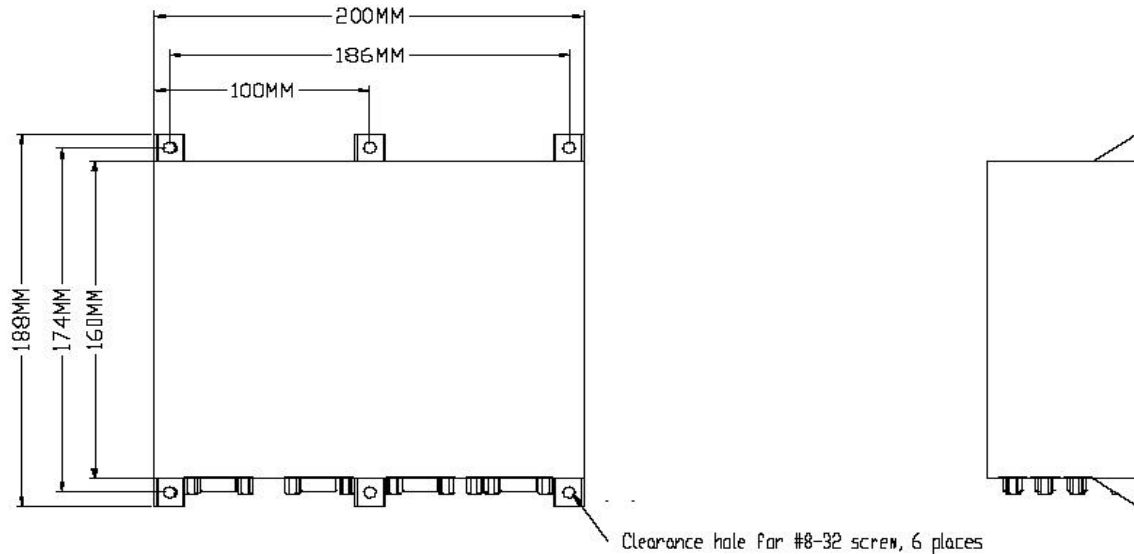
STEREO IMPACT  
IDPU Block Diagram  
D. Curtis Rev B 2001-7-17



## **IDPU Responsibilities**

- **System Design – Dave Curtis**
- **MAG Analog – GSFC / Mario Acuna**
- **MAG Heater – GSFC / Mario Acuna**
- **LVPS – Peter Berg**
- **Data Controller Board, FPGAs – Elf / Dorothy Gordon**
- **STE Detector Interface Board – Steve McBride**
- **Mechanical Design – Heath Bersch**
- **Flight Software – Dave Curtis**
- **EGSE:**
  - **IDPU Simulator Hardware – Elf / Dorothy Gordon**
  - **IDPU Simulator Top Level Software – Mike Hashii**
  - **IDPU Simulator SWEA, STE, MAG Software – Mike Hashii**
  - **IDPU Simulator PLASTIC Software – UNH**
  - **IDPU Simulator SEP Software - Caltech**
  - **Command & Display GSE – Mike Hashii**
  - **SWEA, STE, MAG Science Displays – Mike Hashii**
  - **SEP Science Displays – Caltech**
  - **PLASTIC Science Displays - UNH**

## IDPU



MATERIAL		U.S. STL. THREE ANGLE PROJECTIONS		SPACE SCIENCES LABORATORY		
FINISH				UNIVERSITY OF CALIFORNIA, BERKELEY 94720		
AS MACHINED		STANDARD INTERNAL TOLERANCES		(510) 642-7297 FAX: (510) 643-8302		
DRAWN BY		DIA = .001" ANGLES SURFACES		TITLE		
HEATH BERSCH 6-28-01		DIA X = .012" 0.5		STEREO IMPACT IDPU		
APPROVED		SCALE		DRAWING NO		REVISION
NS/TR		NOT TO SCALE		ICD DRAWING		B
						SHEET

## IDPU Status

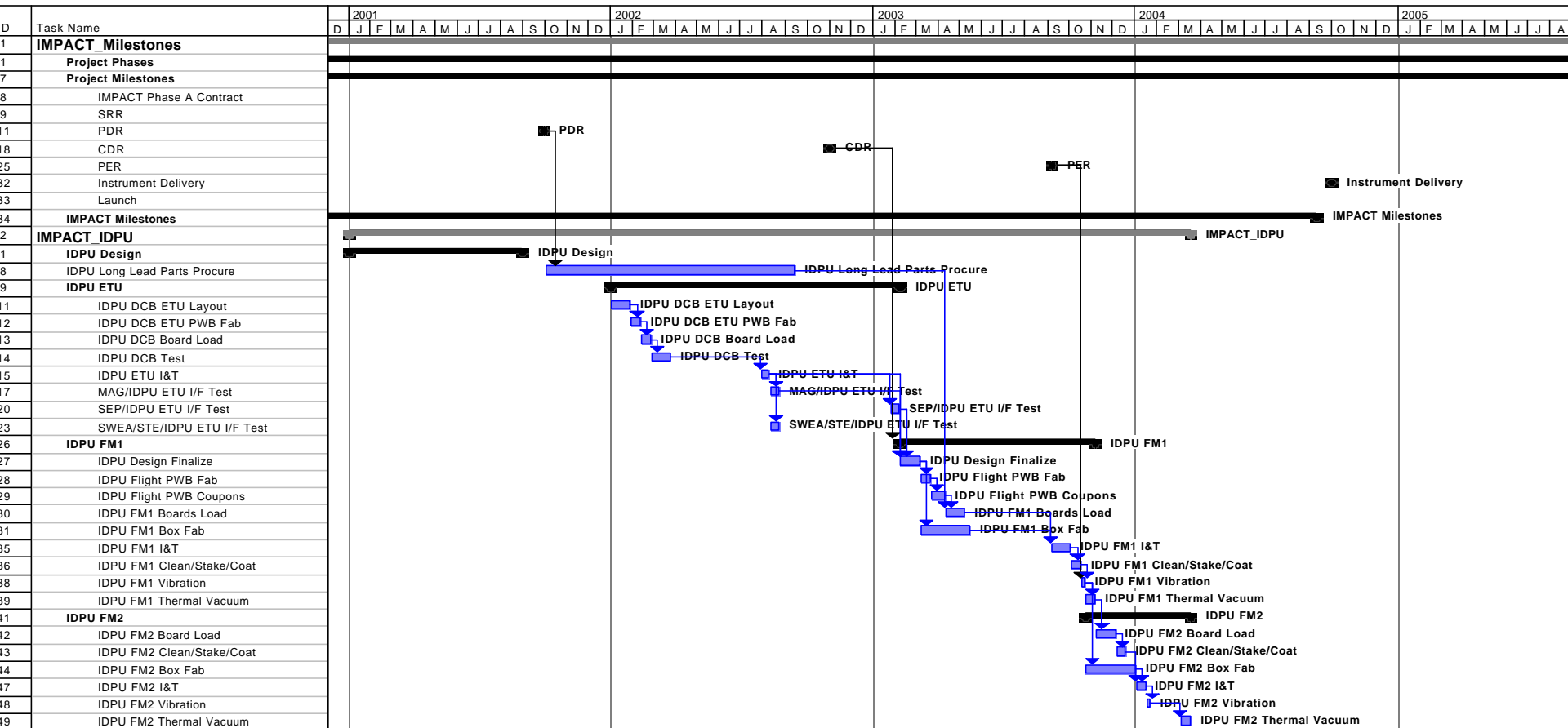
- **IDPU Specification on line at:**
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Specifications/IDPUSpec\\_C.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Specifications/IDPUSpec_C.pdf)
- **IDPU Flight Software Requirements are on-line at:**
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Specifications/IDPUSoftwareRequirements\\_B.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Specifications/IDPUSoftwareRequirements_B.pdf)
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Specifications/PLASTIC\\_Software\\_rec.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Specifications/PLASTIC_Software_rec.pdf)
- **Serial Interface Breadboard to verify interface tested**
- **IDPU Simulator GSE is very similar to the IDPU Data Controller Board (DCB), giving us some early experience with the processor and serial interfaces**
- **Work on ETU DCB to start in January, after the IDPU Simulator is completed**
- **Recent changes to IDPU Memory**
  - Changed addressing to increase maximum code size from 32K to 64K
  - Increased RAM from 2M to 3M due to increased PLASTIC buffering requirements and to allow burst system to make better use of enhanced bitrate
- **Flight Software Requirements / Design to be presented later today**
- **Flight Software coding has started**
  - low level routines to aid in design trades

## **Non-Standard Parts**

- **UCB is working on up-screening a number of non-standard parts, including:**
  - **LT1353C quad opamp**
  - **AD8005A current feedback opamp**
  - **CA3080A transconductance opamp**
  - **MAX987 comparator**
  - **AD7664 16-bit ADC**
  - **LT1599 16-bit DAC**
  - **LT1877 regulator**
  - **MMBT3904, MMBT3906, MMBTH81 surface mount transistors**
  - **LM2672N-ADJ adjustable zener**
- **SCD/Travelers for screening these parts are approved by Project Parts:**
  - **LT1353C, JANTXV1N6642, JANTXV2N2222A, JANTXV2N2907, 54AC14DMQB, 54AC74DMQB and LM193AH/883**
  - **Some only require PIND, some require more extensive testing**
  - **More SCD/Travelers are in the works**
  - **Screening will start shortly to provide time to recover from problems**
  - **LT1024 being up-screened by CESR**
- **Screening costs are very high (up to \$30K a part type)**
- **Some of these parts will be used by other IMPACT team members and non-IMPACT STEREO instrumenters**
  - **Method of sharing screening costs has not been fully worked out**



## IDPU Schedule



# Ground Support Equipment

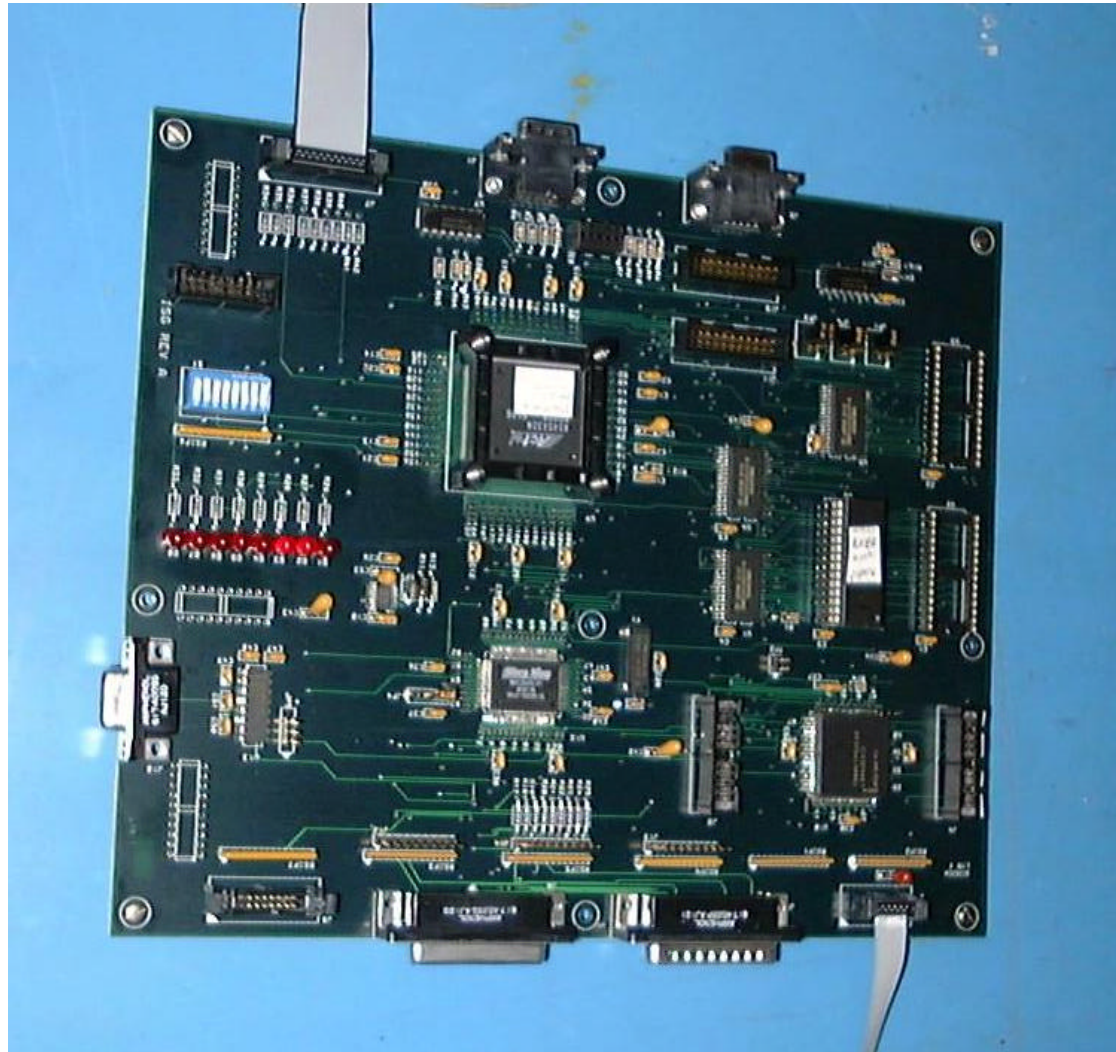
## **IDPU Simulator GSE**

- **Developed at UCB (with Elf)**
- **Provides ability to test instruments in the absence of the IDPU and verify the instrument / IDPU interface**
- **Consists of a black box (IDPU Simulator GSE, ISG) plus a PC**
- **ISG includes most of the functionality of the Data Controller Board, minus the 1553 interface**
  - **Includes a commercial 80C196 processor, FPGA with an early version of the flight DCB FPGA**
  - **Interfaces with PC via the printer port**
  - **Designed to IDPU Simulator Specification Document by Elf (who also designs the DCB)**
  - **Can be used as a test bed for IDPU software**
  - **Has an added Instrument Simulator feature which works the serial instrument interface in the reverse direction for IDPU testing**
- **PC software shall be developed at UCB**
  - **Based on LabWindows CVI development system**
  - **HESSI GSE heritage; STOL-like scripted command system**
  - **Science display modules written at the instrument home institution**

## **IDPU Simulator Status**

- **IDPU Simulator Hardware in test**
- **IDPU Simulator Software in work at SSL**
  - Essential features have been extracted from HESSI GSE
  - Interface to hardware in test
- **Simulator needs to be adapted to the requirements of the Instrument**
  - Display & Command requirements
  - UCB to write the MAG and SWEA/STE versions
  - Caltech to write the SEP version, UNH to write the PLASTIC version
    - UCB can provide MAG version as a template
    - Since the IDPU is a bent pipe for SEP, SEP display requirements will be similar for the IDPU Simulator GSE and the later SEP Science displays;
    - We should look into making a version of the IDPU Simulator GSE that connects to the SEP Science Display GSE used at the Suite level
  - Simulators should be available early next year

## IDPU Simulator Electronics



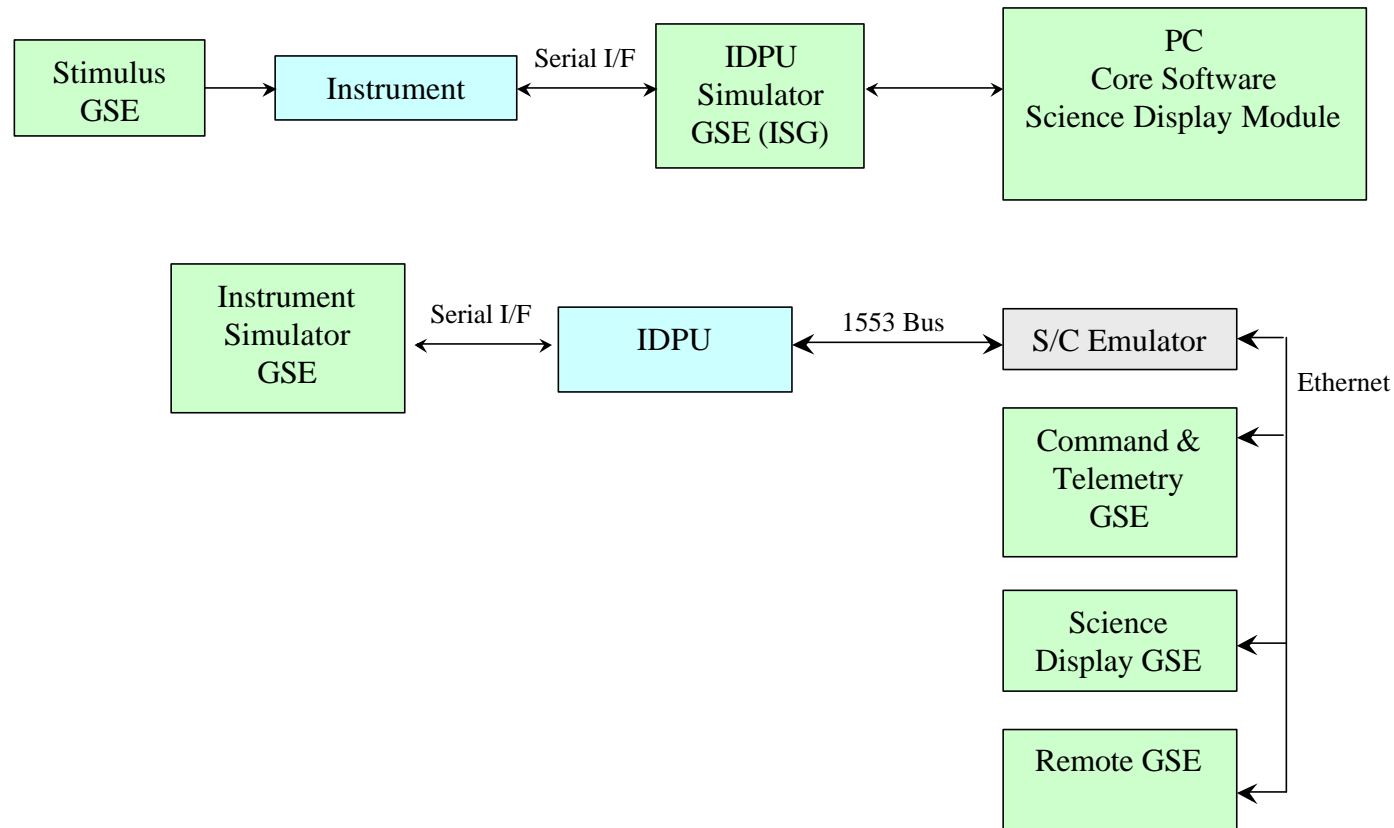
## **Command & Telemetry GSE**

- **Works with Spacecraft Emulator at suite I&T level**
- **Works with MOC at Spacecraft I&T level**
- **Runs commands and command scripts**
  - **STOL-like language**
- **Remote commanding & display via secure internet connection**
- **Displays housekeeping and instrument status information with limit-checking / alarms**
- **PLASTIC command scripts may be run on the IMPACT C&T GSE, or on a separate C&T GSE running the same software**

## **Science Display GSE**

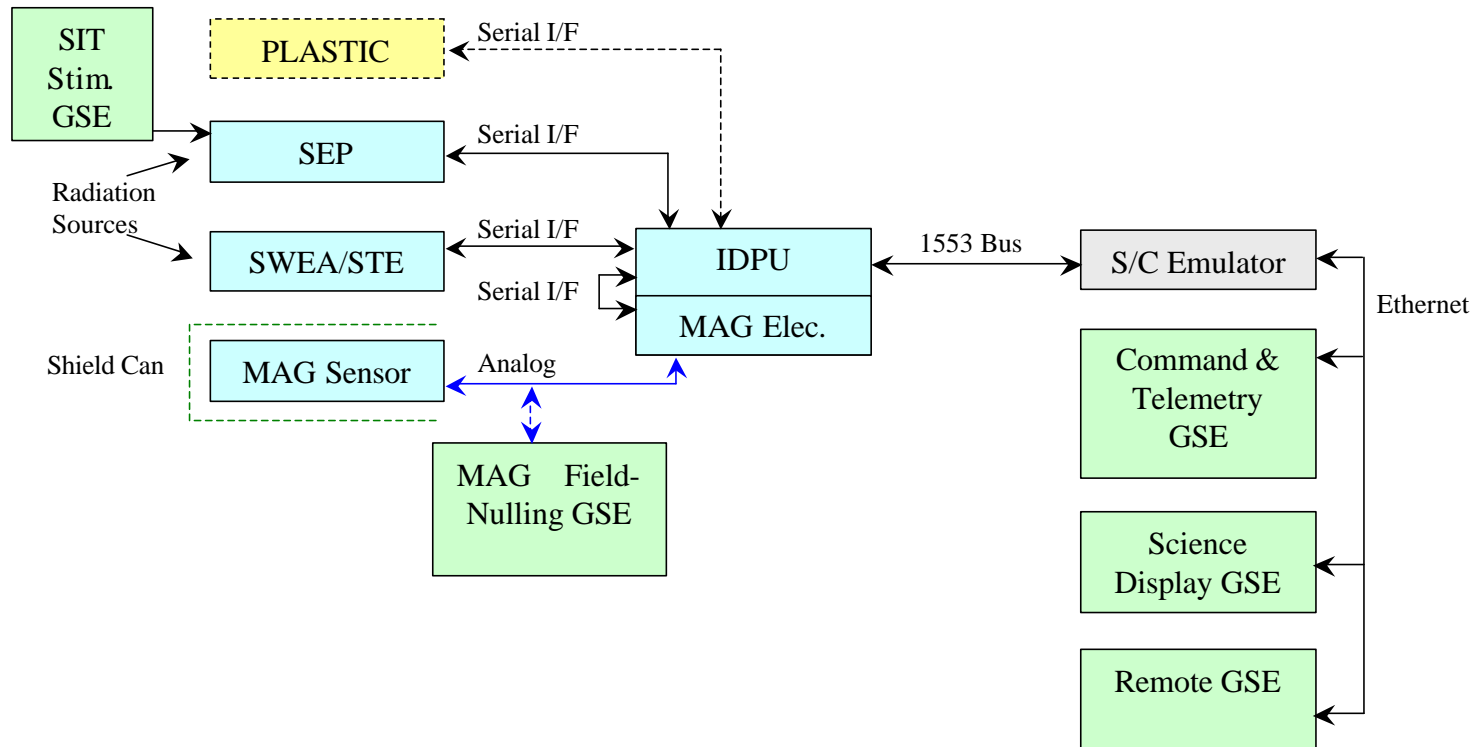
- **Decodes and Displays science data from instruments**
- **Provides adequate information to determine the health and functionality of the instrument in the I&T, Commissioning, and mission environment**
- **Runs on a second workstation (PC) in order to provide more display space, separate science and engineering functions, and improve the reliability of the C&T system**
- **Science Display GSEs will get data from the MOC or Spacecraft Emulator**
- **Science displays shall be developed by the instrument teams:**
  - **MAG, SWEA, STE – UCB / Hashii**
  - **SEP – Caltech**
  - **PLASTIC – UNH**

## Instrument Bench Checkout Configuration

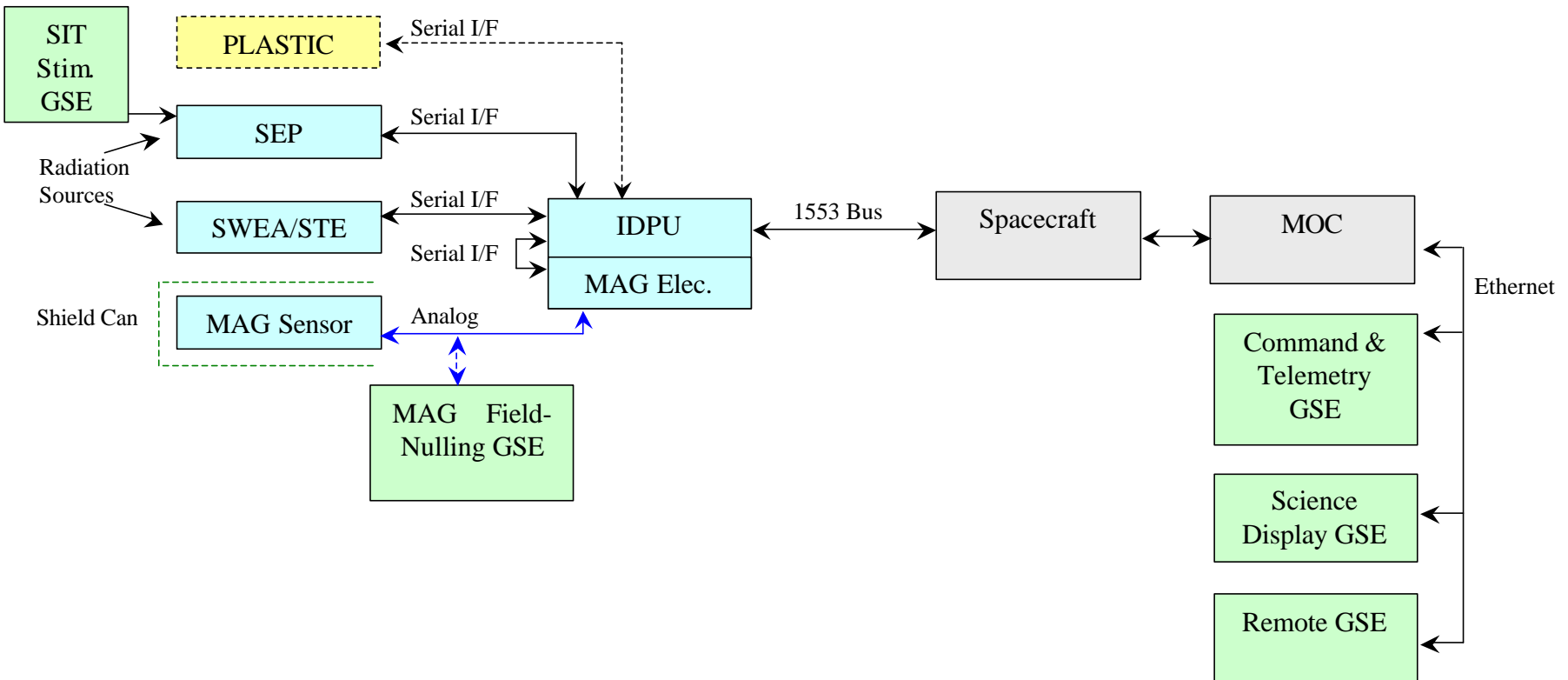




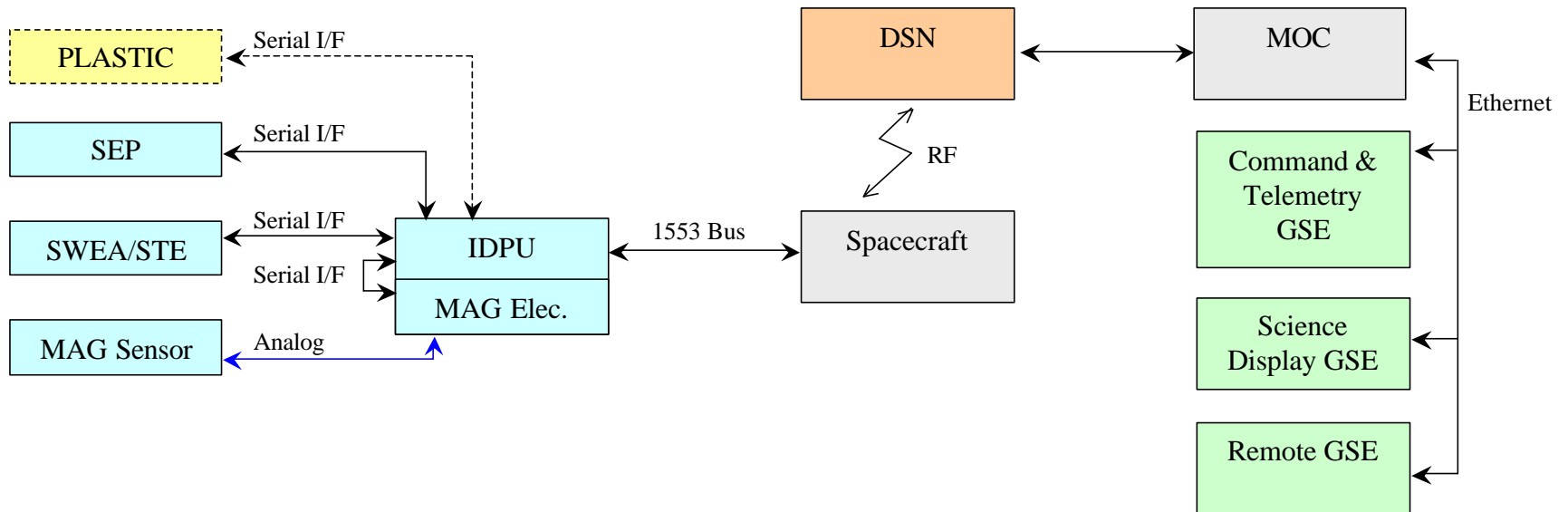
## IMPACT Suite Integration GSE Configuration



## IMPACT Spacecraft Integration GSE Configuration



## IMPACT Mission Operations GSE Configuration



## **Issues**

## Key Documents

- **IMPACT ICD, on APL Forum web site**
- **Environmental Requirements:**
  - <http://sprg.ssl.berkeley.edu/impact/dwc/Project/EnvironmentalSpec7381-9003e.pdf>
- **EMC Requirements:**
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Project/EMC\\_7381-9030d.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Project/EMC_7381-9030d.pdf)
- **Contamination Control Plan:**
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Project/ContaminationControlPlan\\_7381-9006-.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Project/ContaminationControlPlan_7381-9006-.pdf)
- **Import / Export Plan:**
  - <http://sprg.ssl.berkeley.edu/impact/dwc/Project/460-PLAN-0025ImportExport.pdf>
- **IMPACT Performance Assurance Implementation Plan:**
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Plans/STEREO-IMPACT-PAIP\\_D.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Plans/STEREO-IMPACT-PAIP_D.pdf)
- **IMPACT Configuration Management Plan:**
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Plans/IMPACTCMPlan\\_B.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Plans/IMPACTCMPlan_B.pdf)
- **IMPACT Environmental Test Plan:**
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Plans/IMPACTEnvTestPlan\\_A.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Plans/IMPACTEnvTestPlan_A.pdf)
- **IMPACT Performance Specification:**
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Specifications/IMPACTPerformanceSpec\\_F.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Specifications/IMPACTPerformanceSpec_F.pdf)

## **IMPACT PAIP**

- **A new version of the IMPACT PAIP has been negotiated with Project**
  - **Some new requirements for PLASTIC parts qualification**
- **NASA-Funded Institutions need to verify that they can meet the PAIP requirements before we sign off on this PAIP**
- **Major concern with non-NASA Funded Institutions**
  - **PAIP has Appendices describing the PAIP for each institution (except Lindau)**
  - **Project is unhappy about the lack of specifics in these plans**
    - **Some of the referenced documents have a wide range of applications, and there is sometimes no reference to what level is to be used**
  - **Signed-off LOAs state that NASA will provide “Performance Assurance Requirements”**
  - **Can we use GGS Wind 3DP PAIPs for CESR, ESTEC?**

## IMPACT ICD

- **IMPACT ICD has been signed off as of PDR**
  - Under Project change control
- **IMPACT Team needs to review the document, verify the document is correct, and help fill in any TBDs**
  - Issues found late will be costly to fix
- **We need to decide the refurbishment requirements for the instruments on a case by case basis**
  - **PLASTIC will be refurbished while the spacecraft is shipped to the Cape**
  - **Refurbishment can be argued if there is risk of damage to the instrument during I&T which cannot adequately be verified except by an internal inspection or a test that cannot be performed on the spacecraft**
  - **Instruments with MCPs are often refurbished after environmental tests to verify that the detectors were not contaminated, and are completely functional**
  - **This decision is independent of the need to remove an instrument if it has trouble during I&T.**
- **We need a time interval for how long instruments can safely look at the sun**
  - **Kiel has provided a number for SEPT**

## EMC / Grounding

- **STEREO has a severely constrained EMC due to the sensitivity of the SWAVES Instrument Radio receivers**
  - Similar to Wind requirements
  - Sensitive to exposed conductors with millivolts of noise on them
- **Highlights of the EMC control plan are:**
  - Instrument chassis is bonded to spacecraft chassis with less than 5 milli-ohms impedance
  - No exposed harness – all will be over-wrapped continuously with shielding
    - Problem for externally-mounted components such as heaters
  - Strict conducted emissions requirements placed on converters
    - Must run at a multiple of 50KHz, crystal-controlled
  - Signal ground / Secondary Ground / Chassis Ground connected together in each box
    - Helps us keep differential chassis ground / signal ground noise out of our detector front-ends
  - Remotely powered units (such as SEPT) should have isolated converters, connected to signal ground only at the load end
    - Avoids current loops through chassis, which are primarily a concern to MAG
    - We are working on the resource requirements to meet this requirement for SEPT
    - We will probably submit a waiver request for STE-U and SIT
      - The current for STE-U is small
      - SIT is physically close to SEP Central, where the LVPS is



## Telemetry

- Our telemetry allocation has been raised from 2100bps to about 3300bps
- Agreed sample intervals: 1,2,10,30 sec, 1,5,15,30,60 minutes
- Proposed Allocations:
  - SEP Proposal from Rick Cook, using integral numbers of packets per minute
    - Need corresponding new data products & rates
  - MAG sample rate goes from 4Hz to 8Hz
  - STE goes to 8q x 16E @ 10 sec (was 16 sec)
  - SWEA goes to:
    - Moments: 13 @ 2 sec
    - PAD: 12q x 7E @ 10 sec (was 16 sec)
    - 3D: 80W x 16E @ 30 sec (was 60 sec)
  - Remainder goes into Burst playback
- There may be more bits available at CDR

	Was	Proposed
<b>SEP:</b>		
SEPT	120	70
SIT	240	418
LET	320	557
HET	120	209
<b>Boom:</b>		
MAG	192	384
SWEA	394	513
STE	64	102
Burst	486	845
<b>Overhead</b>	164	203
<b>Total</b>	2100	3300

## Beacon Telemetry

- No longer share a beacon packet with PLASTIC
- Allocated one packet per minute (272 bytes, 36bps)
- Proposed Allocation:
  - 60 second resolution except MAG (10 sec)
  - Moments + PAD from SWEA
  - 8E x 2 dir from STE
  - SEP details on next page

	BPS
<b>SEP:</b>	
SEPT	5.87
SIT	3.20
LET	6.13
HET	3.73
<b>Boom:</b>	
MAG	4.80
SWEA	2.13
STE	6.67
<b>Overhead</b>	3.74
<b>Total</b>	36.27

## SEP Beacon Data Detail

SEP		(MeV or MeV/nuc)		Geometry		
Sensor	Species	E1	E2	Factor (cm2sr)	Number of Directions	bps
SIT	He	0.02	0.04	0.3	1	0.27
		0.08	0.16	0.3	1	0.27
		0.32	0.64	0.3	1	0.27
		1.29	2.56	0.3	1	0.27
	CNO	0.02	0.04	0.3	1	0.27
		0.08	0.16	0.3	1	0.27
		0.32	0.64	0.3	1	0.27
		1.29	2.56	0.3	1	0.27
	Fe	0.02	0.04	0.3	1	0.27
		0.08	0.16	0.3	1	0.27
		0.32	0.64	0.3	1	0.27
		1.29	2.56	0.3	1	0.27
					Sum SIT	3.20
SEPT	Electrons	0.02	0.05	0.2	4 separate	1.07
		0.05	0.1	0.2	4 summed	0.27
		0.1	0.2	0.2	4 summed	0.27
		0.2	0.4	0.2	4 separate	1.07
	Ions (mostly protons)	0.02	0.1	0.24	4 separate	1.07
		0.1	0.5	0.24	4 summed	0.27
		0.5	2.5	0.24	4 summed	0.27
		2.5	7	0.24	4 separate	1.07
SEPT Status (1 bit/channel; 20 channels)						0.53
Sum SEPT						5.87

SEP		(MeV or MeV/nuc)		Geometry			
Sensor	Species	E1	E2	Factor (cm2sr)	Number of Directions	bps	
LET	Protons	1.4	3	4.5	2	0.53	
		3	6	4.5	1	0.27	
	Helium	1.7	3	4.5	2	0.53	
		3	6	4.5	1	0.27	
	3He	6	13	4.5	2	0.53	
		1.7	3.3	4.5	1	0.27	
	CNO	3.3	13	4.5	1	0.27	
		3	6	4.5	1	0.27	
	Fe	6	13	4.5	1	0.27	
		13	30	4.5	1	0.27	
		3	6	4.5	1	0.27	
	Livetime	6	13	4.5	1	0.27	
		13	30	4.5	1	0.27	
		30	50	4.5	1	0.27	
							0.27
	H/He-Effic						0.27
	Z-Effic						0.27
	L1A-th						0.27
	L1B-th						0.27
L2L2th						0.27	
Sum LET						6.13	
HET	Electrons	1	4	0.5	1	0.27	
		13	20	0.7	1	0.27	
		20	40	0.7	1	0.27	
	He	50	100	0.7	1	0.27	
		13	20	0.7	1	0.27	
		20	40	0.7	1	0.27	
	CNO	50	100	0.7	1	0.27	
		30	50	0.7	1	0.27	
		50	100	0.7	1	0.27	
	Fe	50	100	0.7	1	0.27	
						0.27	
						0.27	
	Livetime						0.27
Stop Effic.						0.27	
Pen. Effic.						0.27	
HET status						0.27	
Sum HET						3.73	
SEP Status						0.27	
SEP Total						19.20	

## Contamination Control

- The spacecraft has strict contamination control requirements mostly due to the SECCHI instrument
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Project/ContaminationControlPlan\\_7381-9006-.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Project/ContaminationControlPlan_7381-9006-.pdf)
- Dust and Non-volatile residue are carefully controlled; volatiles are of less concern to SECCHI; to the extent that we are concerned with volatiles killing our detectors, we need to continue to remind APL of our concerns
  - Tycho is doing a good job of pressing this issue
  - Quantitative requirements would be easier for the system to handle
- Instruments must be delivered to the spacecraft clean and baked out
  - External cleanliness will be verified at APL, probably class 300
  - Internal cleanliness must be imposed by the teams procedurally, or else the boxes must be sealed with a filtered vent to prevent dust from getting out
  - Bakeout must be verified by QCM measurement to ensure outgassing meets requirements
  - Failure to meet outgassing requirements will result in a long bakeout or possibly even dismantlement to remove the offending material

## Environmental Requirements

- **Environmental Requirements (except EMC) are not very unusual**
  - <http://sprg.ssl.berkeley.edu/impact/dwc/Project/EnvironmentalSpec7381-9003e.pdf>
  - Note that only Section 3 applies to instruments
- **EMC Requirements are stringent as noted above**
  - [http://sprg.ssl.berkeley.edu/impact/dwc/Project/EMC\\_7381-9030d.pdf](http://sprg.ssl.berkeley.edu/impact/dwc/Project/EMC_7381-9030d.pdf)
- **Parts must meet an 8KRad total dose and 80 MeVcm<sup>2</sup>/mg Latchup Level**
- **Instrument-level tests on BOTH flight units include:**
  - Thermal Vacuum
  - Vibration: Shock, Sine Sweep, Random (Acoustic as required)
  - Mass Properties
  - EMC
  - Magnetics (Mario's test setup)
- **We will be doing EMC as an integrated suite. The remaining tests are mostly at the box level**
- **The Electrostatic requirement is that exterior surfaces must be less than 10<sup>8</sup>W/square**
  - Any non-conducting exterior surfaces must be identified and tracked at the system level
- **Magnetic Materials shall also be tracked at the system level**

## **Schedule**

- **There is concern that there is insufficient slack in the system schedule**
- **Project has moved up our delivery date to improve the slack in I&T, and would like to move us up some more**
- **SEP is currently on the “Critical Path” for the mission**
  - This is due primarily to key personnel issues at Caltech and GSFC
- **We need to work to improve slack**
  - There is probably some “hidden” slack buried in tasks with generous durations
  - It is human nature to use up all the available schedule, so the buried slack will get eaten up, leaving nothing for problems encountered late in the flow
  - Slack must be managed like mass, power, and dollars. Managers must maintain adequate margins. Like other resources, schedule margin must be guarded closely.
  - There is some resistance to divulging all our slack for fear that it will get taken away from us by Project (witness the 4 weeks we recently lost). This must be balanced by the danger that inadequate slack will be detrimental to getting the mission confirmed.
- **One way to improve our schedule is to skip final integration at UCB, and have SEP deliver directly to APL. The team needs to consider this possibility.**
- **Project wants regular updates to our schedule so they can track progress**
  - We first need to firm up our schedules so they match up and show adequate margin
  - We then need to report on schedule status compared to this baseline monthly

## **Miscellaneous Issues**

- **We need to maintain our Parts & Materials Lists, preferably on a common format**
  - Do I have HET & SIT Digital parts lists?
  - Parts lists must be approved by Project
- **We need to develop a more detailed data analysis plan**
  - Data formats, data products, etc.
  - We should set up a committee of concerned parties and have a kick-off telecon early next year
- **Project would like to know what our ETU plans are:**
  - What their fidelity is, what interface testing is done, and where they are maintained.
- **We need to complete our RFA responses. Project has some comments to our initial responses which team members need to respond to.**